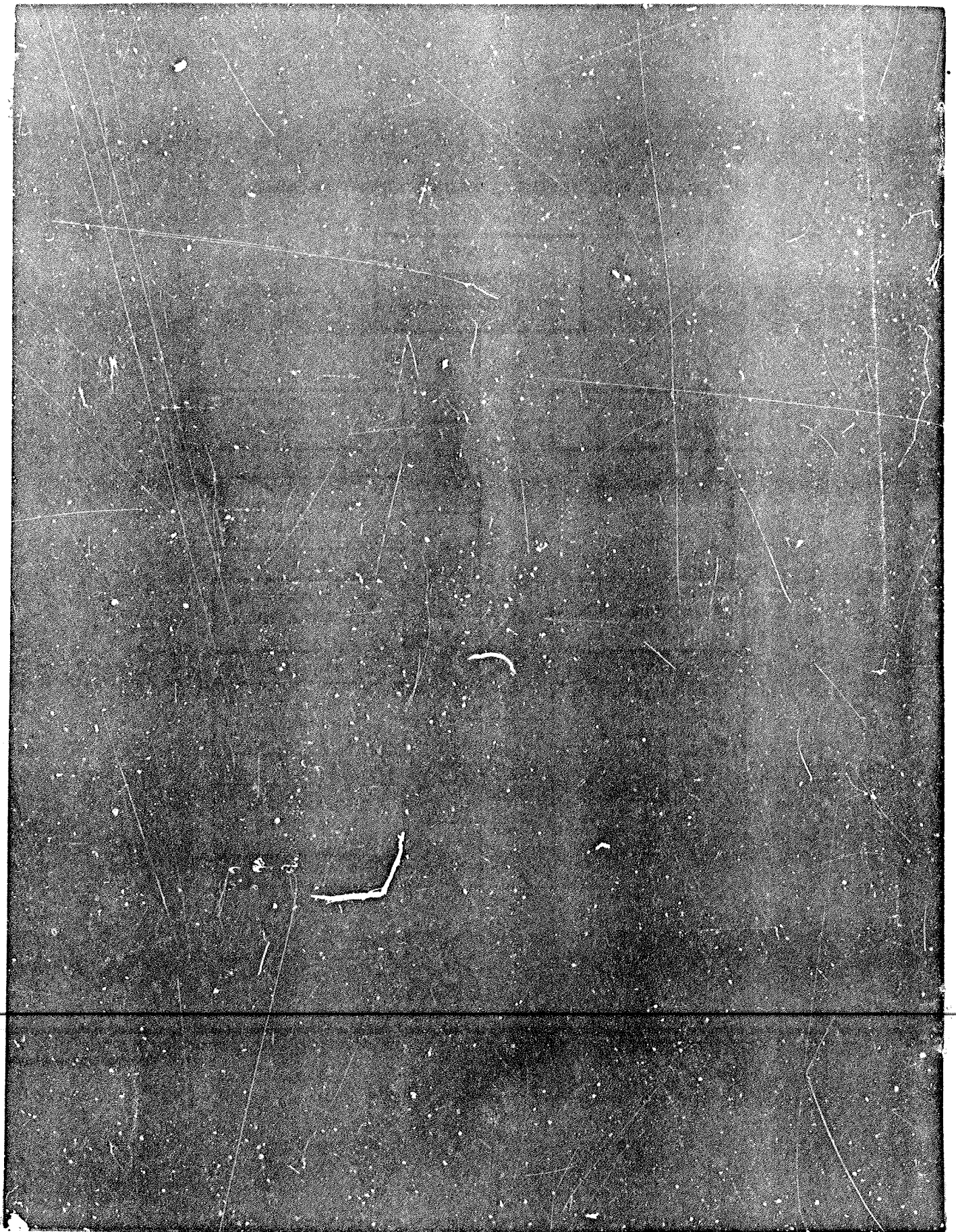


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coefficients. The presence of asymmetry (camber) also is accounted for. Predictions of towing configurations are presented. It is shown that small angles of attack (.15 degrees) can result in kite angles up to 60 degrees and catenary depth losses of 10 to 15 percent. Remarkably small camber (.1 percent of chord) is sufficient to cause significant kiting. Stability criteria are developed based on an analysis of the torsional buckling problems associated with the structural rigidities. The results have important implications in the design of towable strength members and fairings.

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## NOTATION

$C$	Chord length
$C_d$	Drag coefficient
$C_l$	Lift coefficient
$C_m$	Moment coefficient
$c_o$	Camber
$d$	Drag per unit span
$\bar{e}_i$	Unit vector in $i^{\text{th}}$ direction
$f_{s,n}$	Drag loading functions
$f = (f_s, f_k, f_n)$	Hydrodynamic force
$F = (T, F_k, F_n)$	Structural force
$K = (K_k, K_n)$	Curvature
$l$	Lift per unit span
$L$	Cable length
$m = (m_s, m_k, m_n)$	Hydrodynamic moment
$M = (M_s, M_k, M_n)$	Structural moment
$s$	Cable arc length
$(s, k, n)$	Curvilinear coordinates
$(x, y, z)$	Cartesian coordinates
$(X, Y, Z)$	Cable geometry functions
$V$	Free stream velocity
$\alpha$	Angle of attack
$\epsilon, \delta$	Small parameter
$\theta$	Kite angle
$(\xi, \eta)$	Cable cross section coordinates

$\rho$  Fluid mass density  
 $\phi$  Trail angle  
 $\tau$  Twist



## ABSTRACT

A theory for faired towable kiting is presented. Geometric relationships and equilibrium equations are derived based on a thin elastic rod-high aspect ratio hydrofoil analytical model. The fairing section is characterized by torsional and flexural rigidities and hydrodynamic lift, drag, and moment coefficients. The presence of asymmetry (camber) also is accounted for. Predictions of towing configurations are presented. It is shown that small angles of attack (.15 degrees) can result in kite angles up to 60 degrees and catenary depth losses of 10 to 15 percent. Remarkably small camber (.1 percent of chord) is sufficient to cause significant kiting. Stability criteria are developed based on an analysis of the torsional buckling problems associated with the structural rigidities. The results have important implications in the design of towable strength members and fairings.

## ADMINISTRATIVE INFORMATION

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## INTRODUCTION

The use of cable fairing in underwater cable towed systems has been successful in reducing hydrodynamic drag limitations on the speed/depth performance. Generally the fairing is a symmetric airfoil shape configured around the primary tension members in a free swivelling or integrally bonded construction. Under these circumstances, the cable may no longer act as a "taut string" for which tension and drag are the principal forces determining the catenary or towing configuration. Instead, the fairing is a lifting surface capable of developing large hydrodynamic side forces. Furthermore, the fairing structural rigidities introduce flexural and torsional loading in the towing catenary. These properties have important effects on both the steady state tracking characteristics and dynamic response of the towed system.

Faired cable is subject to a type of lateral instability, usually referred to as kiting. Kiting is the tendency of the cable to displace and remain out of the intended towing plane (usually the gravity-tow velocity plane). Previous analytical investigations, based on small perturbation/energy integral methods,<sup>1,2</sup> resulted in criteria for stable or kite-free

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<sup>1</sup>Abkowitz, M.A., "The Stability of a Faired Cable of a Tethered System in its Fundamental Mode," Joseph Kaye and Co., Inc., Report 73 (Sep 1967).

<sup>2</sup>Hegemoir, G.A., "Divergence Criteria for a Faired Towable in a Subcavitating Flow," University of California, San Diego (Jun 1968).

towing in terms of cable physical properties and towing velocity and tension. Fairing asymmetries (camber) are not accounted for in these theories. A number of digital computer programs have been developed to compute general three dimensional towing configurations with prescribed camber distributions.<sup>3,4</sup> However, numerical integration is difficult when torsional and flexural loading is introduced and solutions which account for these effects are not yet available.

In this report a new treatment of towcable kiting theory is presented. A thin elastic-rod, high aspect ratio hydrofoil analytical model is used to determine the importance of both camber and destabilizing structural loading. The assumption that the cable cross section length is much smaller than the total scope and radius of curvature (thin rod) permits the use of small strain beam theory. The assumption that the spanwise variation in hydrodynamic loading is small permits the use of two-dimensional lifting surface theory. Both of these approximations are particularly appropriate to integrated towlines, which are used in high speed/deep depth towing where kiting is most severe.

The governing equations of force equilibrium and geometric compatibility are derived with particular emphasis on the role of the fairing angle of attack. By systematically ordering terms, it is deduced that the curvatures and towing catenary may be computed considering only tension and lift/drag loading. This is illustrated for several selected hydrodynamic loading functions.

The importance of the amplitude and distribution of camber is assessed and identified as a primary cause of kiting. It is shown that the chordwise locations of both the tension and hydrodynamic centers are crucial to the degree of camber induced kiting.

Further analysis considers the structural instability of a fairing in a towing catenary. Solutions in terms of known mathematical functions are derived for a selected set of cable end constraints. It is shown that under certain circumstances, torsional and flexural rigidity can substantially affect the spanwise twist distribution and resulting hydrodynamic side loading. This theory may be extended to more general cases. Conclusions are drawn which have important implications in the design of fairing and cable strength members.

#### GEOMETRIC RELATIONSHIPS

Consider a cable-body towed system advancing at constant speed,  $V$ , through otherwise undisturbed water. As a result of hydrodynamic, structural, and weight loading the cable forms a three-dimensional continuous arc from the body to the towing platform. The spatial configuration of the cable is defined in terms of a cartesian coordinate system  $(x,y,z)$  moving

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<sup>3</sup>Dillon, D.B., "The Configuration and Loading of a Torsionally Elastic Faired Cable," Hydrospace Challenger, Inc., TR-4557-0001 (Oct 1973).

<sup>4</sup>Wang, H.T., "A FORTRAN IV Program for the Three Dimensional Steady State Configuration of Extensible Flexible Cable Systems," Naval Ship Research and Development Center, Report 4384 (Sep 1974).

with the cable with origin at the body. The x-axis is parallel to the free stream, positive into the flow; y is directed opposite to the gravity vector; and z completes the right hand system. If s denotes arc length measured from the origin, then the geometry of the cable arc can be specified parametrically as  $[X(s), Y(s), Z(s)]$  in terms of the trail angle,  $\phi(s)$ , and the kite angle,  $\theta(s)$ , defined in Figure 1.

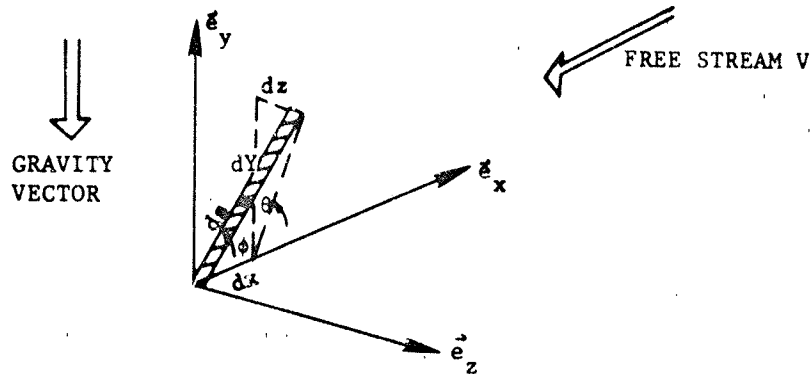


Figure 1 - Geometry of Cable Arc Element

The displacement-angle relationships are

$$\frac{dX}{ds} = \cos \phi \quad (1)$$

$$\frac{dY}{ds} = \sin \phi \cos \theta \quad (2)$$

$$\frac{dZ}{ds} = \sin \phi \sin \theta \quad (3)$$

A unit vector,  $\vec{e}_s$ , tangent to the arc is given by

$$\vec{e}_s = \vec{e}_x \cos \phi + \vec{e}_y \sin \phi \cos \theta + \vec{e}_z \sin \phi \sin \theta \quad (4)$$

and the curvature of the arc,  $\vec{K}$ , is

$$\begin{aligned} \vec{K} = \frac{d\vec{e}_s}{ds} = & \vec{e}_x \left( -\sin \phi \frac{d\phi}{ds} \right) + \vec{e}_y \left( \cos \phi \cos \theta \frac{d\phi}{ds} - \sin \phi \sin \theta \frac{d\theta}{ds} \right) \\ & + \vec{e}_z \left( \cos \theta \sin \phi \frac{d\theta}{ds} + \cos \phi \sin \theta \frac{d\phi}{ds} \right) \end{aligned} \quad (5)$$

It will be convenient to obtain relationships in terms of curvilinear coordinates  $(s, k, n)$ , where  $n$  is normal to the cable arc in the  $\phi$  plane and  $k$  is normal to the  $\phi$  plane. Observe that drag forces act only in the  $\phi$  plane while hydrodynamic lift forces act in the  $k$  direction. In the absence of kiting, the  $\phi$  plane is the gravity-tow velocity plane everywhere along the cable. Unit vectors in the  $k$  and  $n$  directions are given by

$$\vec{e}_k = \frac{\vec{e}_x \times \vec{e}_s}{\sin \phi} = \vec{e}_y (-\sin \theta) + \vec{e}_z \cos \theta$$

$$\vec{e}_n = \vec{e}_s \times \vec{e}_k = \vec{e}_x \sin \phi + \vec{e}_y (-\cos \phi \cos \theta) + \vec{e}_z (-\cos \phi \sin \theta)$$

and the corresponding components of curvature  $K_k$  and  $K_n$  by

$$K_k = \vec{e}_k \cdot \vec{K} = \sin \phi \frac{d\theta}{ds} \quad (6)$$

$$K_n = \vec{e}_n \cdot \vec{K} = \frac{-d\phi}{ds} \quad (7)$$

The twist (change in rotation about the  $s$ -axis per unit length  $s$ ) at any location along the cable is related to the kite angle and,  $\alpha$ , the local angle of attack between the fairing chordline and the flow ( $\phi$  plane), as illustrated in Figure 2.

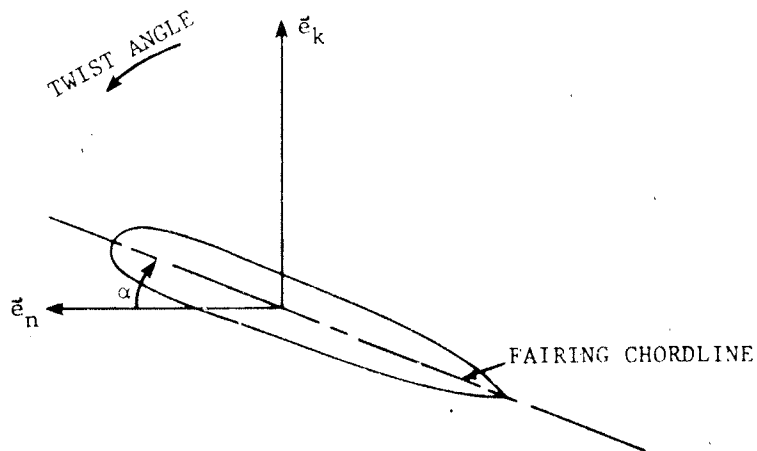


Figure 2 - Fairing Section Angle of Attack

The total twist,  $\tau$ , may be written as

$$\begin{aligned}\tau &= \frac{-d\alpha}{ds} + \vec{e}_n \cdot \frac{d\vec{e}_k}{ds} \\ &= \frac{-d\alpha}{ds} + \cos \phi \frac{d\theta}{ds}\end{aligned}\quad (8)$$

Note that for a vertical cable ( $\phi = 90^\circ$ ) or a cable without kite ( $\theta = 0$ ) the twist angle is simply  $-\alpha$ .

#### FORCE AND MOMENT EQUATIONS

The steady state loading on the cable consists of hydrostatic, hydrodynamic, and structural forces and moments. To simplify the analysis, hydrostatic loading is neglected. Thus the cable is assumed to be neutrally bouyant and to have no hydrostatic couple exerted on the fairing.

If  $\vec{F}(s)$  and  $\vec{M}(s)$  are the structural force and moment at any cross section along the cable, then by considering a small element,  $\Delta s$ , and taking the limit as  $\Delta s \rightarrow 0$ , the equations of static equilibrium may be written as

$$\frac{d\vec{F}}{ds} + \vec{f} = 0 \quad (9)$$

$$\frac{d\vec{M}}{ds} + \vec{e}_s \times \vec{F} + \vec{m} = 0 \quad (10)$$

where  $\vec{f}(s) = (f_s, f_k, f_n)$  and  $\vec{m}(s) = (m_s, m_k, m_n)$  are the hydrodynamic force and moment per unit length. These are the classical equations for the loading of a thin rod.<sup>5</sup> Assuming that  $\vec{M}$ ,  $\vec{f}$ , and  $\vec{m}$  can be expressed as functions of the cable angles  $\phi$ ,  $\theta$ , and  $\alpha$ , equations (9) and (10) are six ordinary differential equations for the variables  $\vec{F}(s)$ ,  $\phi(s)$ ,  $\alpha(s)$ ,

$$s: \frac{dT}{ds} - F_k K_k - F_n K_n + f_s = 0 \quad (9)(a)$$

$$k: \frac{dF_k}{ds} - F_n K_s + T K_k + f_k = 0 \quad (9)(b)$$

$$n: \frac{dF_n}{ds} + F_k K_s + T K_n + f_n = 0 \quad (9)(c)$$

$$s: \frac{dM_s}{ds} - M_k K_k - M_n K_n + m_s = 0 \quad (10)(a)$$

<sup>5</sup>Hildebrand, F., "Advanced Calculus for Applications," Chapter 6, Prentice-Hall, Inc., New Jersey (1962).

$$k: \frac{dM_k}{ds} - M_n K_s + M_s K_k - F_n + m_k = 0 \quad (10)(b)$$

$$n: \frac{dM_n}{ds} + M_k K_s + M_s K_n + F_k + m_n = 0 \quad (10)(c)$$

where  $\vec{F} = (T, F_k, F_n)$ ,  $\vec{M} = (M_s, M_k, M_n)$ , and  $K_s = \tau + \frac{d\alpha}{ds}$ . Here, the following identities for the derivatives of unit vectors have been used:

$$\frac{d\vec{e}_s}{ds} = \vec{e}_k K_k + \vec{e}_n K_n$$

$$\frac{d\vec{e}_k}{ds} = -\vec{e}_s K_k + \vec{e}_n K_s$$

$$\frac{d\vec{e}_n}{ds} = -\vec{e}_s K_n - \vec{e}_k K_s$$

The hydrodynamic loading is approximated using high aspect ratio hydrofoil theory. Since the fairing chord length,  $C$ , is extremely small compared to the characteristic length of spanwise lift and drag variation, vortex wake induced effects on cable loading are minimal. Thus, the force and moment are functions of the local orientation to the free stream,  $V$ , and the section characteristics  $C_d(\alpha)$ ,  $C_l(\alpha)$ , and  $C_m(\alpha)$ . For small angles of attack,

$$d(\alpha) \sim (C_d)_{\alpha=0} \rho/2 V^2 C \quad (11)$$

$$l(\alpha) \sim \left(\frac{dC_l}{d\alpha}\right)_{\alpha=0} \rho/2 V^2 C \cdot \alpha \quad (12)$$

$$m(\alpha) \sim \left(\frac{dC_m}{d\alpha}\right)_{\alpha=0} \rho/2 V^2 C^2 \cdot \alpha \quad (13)$$

where  $d$ ,  $l$ , and  $m$  denote drag, lift, and moment per unit span length. The flow velocity component normal to the cable,  $V\vec{e}_n$ .  $\vec{e}_n = V \sin \phi$ , is assumed to be the "effective velocity" in producing lift. Thus

$$\vec{f} = [-d\delta_s(\phi), l(\alpha) \sin^2 \phi, -d\delta_n(\phi)] \quad (14)$$

$$\vec{m} = [m(\alpha) \sin^2 \phi, 0, 0] \quad (15)$$

where  $\delta_s(\phi)$  and  $\delta_n(\phi)$  are hydrodynamic drag loading functions. In recent years, theoretical and experimental analyses of these functions have been conducted for a variety of fairing shapes and constructions. Two reviews

of this subject are given in references 6 and 7. For purposes of analyzing towline kiting, several representative functions have been selected. These are introduced later in this report.

Since the section chord is much smaller than the radius of curvature, the cable behaves as a thin rod. Except possibly near attachment points, the strains will be small and it can be assumed that the stress-deformation relationships follow the elastic laws for the materials.<sup>8</sup> Thus, the structural moment,  $\bar{M}$ , can be expressed in terms of curvature and twist by elementary elastic beam theory. For a symmetric integrated towline the properties of the materials and cross section shape needed for the analysis are shown in Figure 3 and Table 1.

In terms of curvatures  $K_1$  and  $K_2$ ,  $\bar{M}$  may be written as

$$\bar{M} = \bar{e}_s \bar{GJ} \tau + \bar{e}_1 [-T \cdot (\xi_T - \xi_s) - \bar{EI}_1 K_2] + \bar{e}_2 \bar{EI}_2 K_1 \quad (16)$$

where  $s$  is chosen to be along the elastic axis (shear center). An equivalent form of equation (16) in  $(s, k, n)$  components, linearized in  $\alpha$ , is

$$\begin{aligned} \bar{M} = \bar{e}_s \bar{GJ} \tau + \bar{e}_k [\alpha K_k (\bar{EI}_2 - \bar{EI}_1) - \bar{EI}_1 K_n - T(\xi_T - \xi_s)] \\ + \bar{e}_n [\bar{EI}_2 K_k - \alpha K_n (\bar{EI}_2 - \bar{EI}_1) + \alpha T \xi_T] \end{aligned} \quad (17)$$

Equations (9), (10), (14), (15), and (17) form the problem for a kiting towline. Subject to appropriate boundary conditions at the towed body ( $s = 0$ ) and the towing platform ( $s = L$ ), the equations can, in principle, be solved for structural loading,  $\bar{F}$ , and configuration  $\phi$ ,  $\theta$ , and  $\alpha$ .

#### KITING WITH A CONSTANT ANGLE OF ATTACK

The six equations (9) and (10) can be reduced to four by solving for the shear forces  $F_n$  and  $F_k$  in (10)(b) and (10)(c) and substituting into (9)(b) and (9)(c). This yields three force equations,

$$\frac{dT}{ds} + f_s = K_n \frac{dM_k}{ds} - K_k \frac{dM_n}{ds} - (\cot \phi) (K_k) (K_n M_n + K_k M_k) \quad (18)$$

<sup>6</sup>Casarella, M. J. & Parsons, M., "Cable Systems Under Hydrodynamic Loading," MTS Journal Vol. 4, No. 4 (Jul-Aug 1970).

<sup>7</sup>Folb, R., "Experimental Determination of Hydrodynamic Loading for Ten Cable Fairing Models," SPD R&D Report 4610 (in progress)

<sup>8</sup>Love, A.E.H., "A Treatise in the Mathematical Theory of Elasticity," Chapter 18, Dover, New York (1944).

TABLE 1

Definitions of Cable Structural Coefficients

Coefficient	Symbol	Formula
Torsional Rigidity	$\overline{GJ}$	$\frac{\int (G) [(\xi - \xi_s)^2 + \eta^2] dA}{\text{AREA}}$
Chordwise Flexural Rigidity	$\overline{EI}_1$	$\frac{\int (E) (\xi - \xi_s)^2 dA}{\text{AREA}}$
Lateral Flexural Rigidity	$\overline{EI}_2$	$\frac{\int (E) (\eta^2) dA}{\text{AREA}}$
Tension Center	$(0, \xi_T)$	$\frac{\int (E) (\xi) dA}{\int (E) dA}$ $\frac{\text{AREA}}{\text{AREA}}$
Shear Modulus	G	-----
Elastic Modulus	E	-----

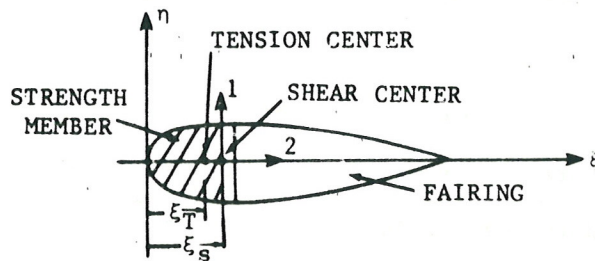


Figure 3 - Fairing Cross Section



$$TK_k + f_k = (\cot \phi)(K_k) \left( \frac{dM_k}{ds} - K_{kn} M \cot \phi + M_{sk} K_k \right) + \frac{d}{ds} \left( \frac{dM_n}{ds} + M_{sn} K_n + M_{kn} K_k \cot \phi \right) \quad (19)$$

$$TK_n + f_n = (\cot \phi)(K_k) \left( \frac{dM_n}{ds} + M_{sn} K_n + M_{kn} K_k \cot \phi \right) - \frac{d}{ds} \left( \frac{dM_k}{ds} - K_{kn} M \cot \phi + K_{ks} M_s \right) \quad (20)$$

and the twist moment equation

$$\frac{dM_s}{ds} - M_{kn} K_k - M_{ns} K_n + m_s = 0 \quad (21)$$

To compare magnitudes of the various terms, the tension, T, is scaled by tension at the body,  $T_0$ . An appropriate length scale for s is  $T_0/d$ . It will be seen that this length is a measure of the radius of curvature at the towed body. The non-dimensional force equations (18), (19), and (20) are

$$\frac{dT}{ds} + \epsilon_s = 0(\epsilon) \quad (22)$$

$$T \sin \phi \frac{d\theta}{ds} + \frac{\ell \sin^2 \phi}{d} = 0(\epsilon) \quad (23)$$

$$T \frac{d\phi}{ds} + \epsilon_n = 0(\epsilon) \quad (24)$$

where the expression  $0(\epsilon)$  denotes terms of order  $\frac{\overline{EI}_1 d^2}{T_0^3}$ ,  $\frac{\overline{EI}_2 d^2}{T_0^3}$  and

$\frac{\overline{GJ} d^2}{T_0^3}$ . Generally, for those circumstances in which kiting occurs,

these parameters are several orders of magnitude less than the terms on the left hand sides. Thus, over most of the cable span, the  $0(\epsilon)$  terms can be ignored. This amounts to assuming that structural rigidity is ignored or that the cable is flexible. Except for small regions near the cable ends, this is a valid approximation in determining solutions for  $\phi$ ,  $\theta$ , and T.

Equations (21) and (23) (with right hand side set to zero) are identical to the two-dimensional (planar) towing equations for a flexible cable, provided that the drag loading function assumptions (equations (11)